

**IN THE SPECIFICATION:**

Please replace the paragraph at page 6, lines 27-32, with the following paragraph:

The blade 10 of Figure 1 generally includes an airfoil 12 against which hot combustion gases are directed during operation of the gas turbine engine, and whose surface is therefore subject to severe attack by oxidation, corrosion and erosion. The airfoil 12 is anchored to a turbine disk (not show) with a dovetail 14 formed on a platform 16 of the blade 10. Cooling holes ~~18~~ 18 are present in the airfoil 12 through which bleed air is forced to transfer heat from the blade 10.